

JAXA's Research and Development of Unmanned Aircraft System Technologies for a Disaster Risk Reduction

Japan Aerospace Exploration Agency (JAXA)

Aeronautical Technology Directorate

Daisuke Kubo



Aeronautical field and centers of ATD/JAXA



Taiki Aerospace Research Field
(1000 x 30 m)

Chofu Aerospace Center
(Chofu airport, 800 x 30 m)

Nagoya Flight Research Center
(Nagoya airport, 2740 x 45 m)



Three programs and basic research area

航空環境技術の研究開発プログラム
ECAT
Environment-Conscious Aircraft Technology Program

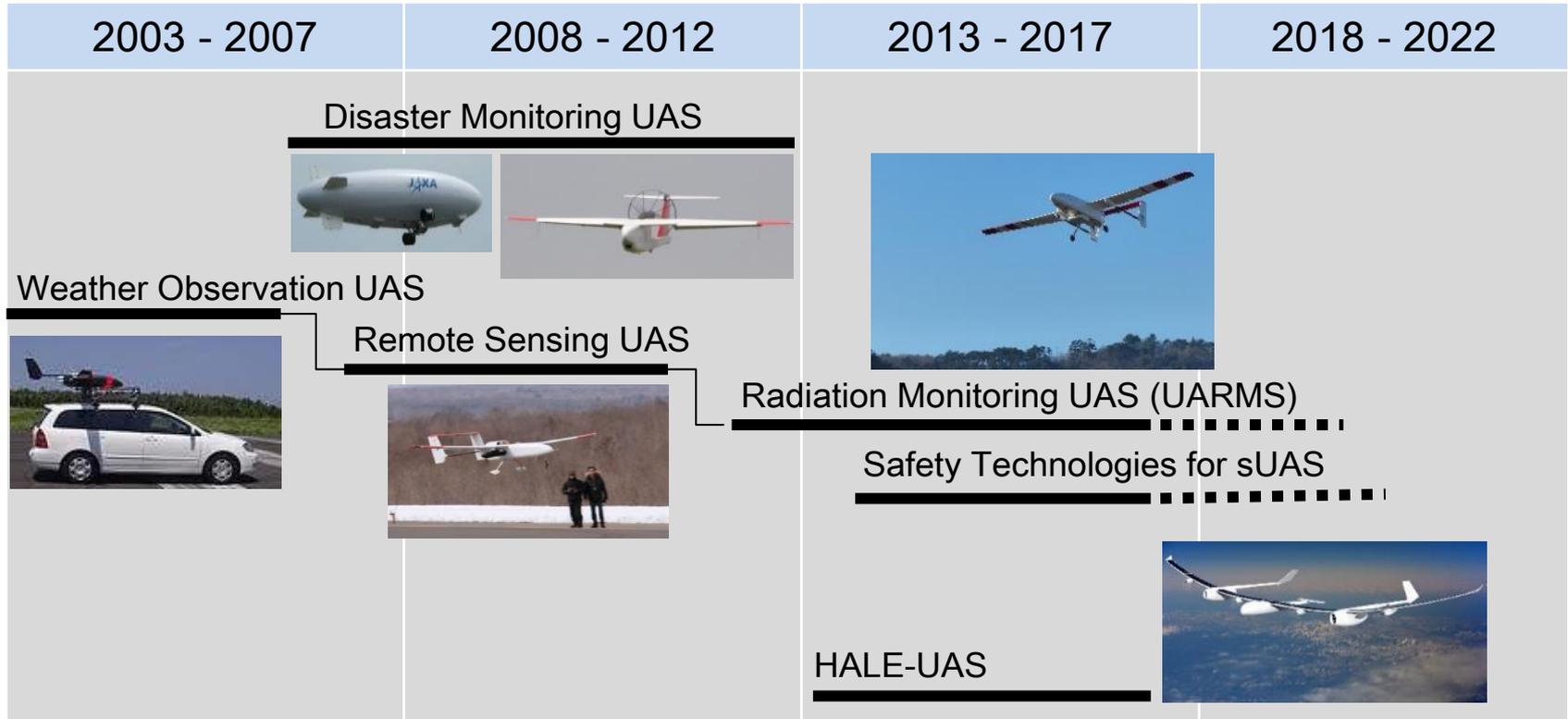
航空安全技術の研究開発プログラム
STAR
Safety Technology for Aviation and Disaster-Relief Program

航空新分野創造プログラム
Sky Frontier
Sky Frontier Program

基礎的・基盤的技術の研究
Science & Basic Tech.
Aeronautical Science & Basic Technology Research

Unmanned Aircraft Systems research and development

Recent R&D related to UAS in JAXA



Unmanned aircrafts for aerospace technology demonstration



Reentry Vehicle Technology
(ALFLEX, HSF)



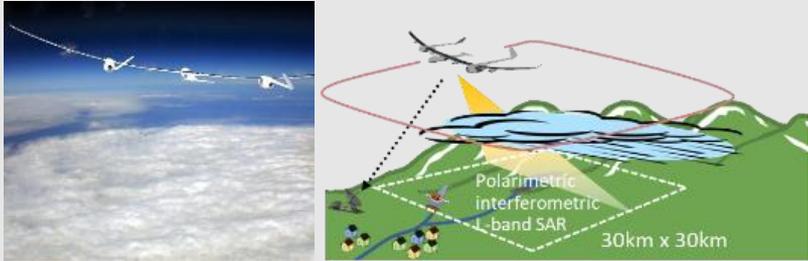
Tiltwing VTOL



Aerodynamics for Supersonic Transport
(NEXST-1, D-SEND)



This talk includes... ongoing 4 UAS R&Ds



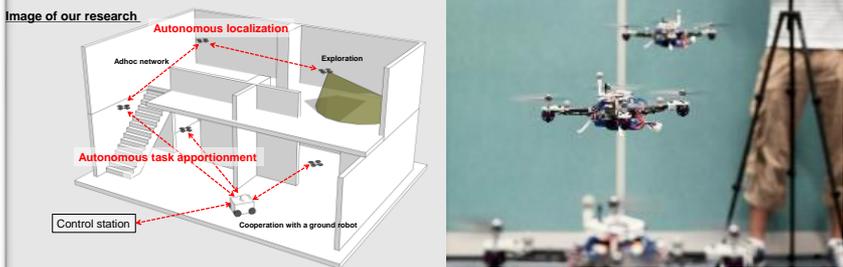
The image shows a satellite view of Earth on the left and a 3D diagram of a UAS on the right. The diagram illustrates a UAS flying over a 30km x 30km area, performing polarimetric interferometric L-band SAR. The SAR beam is shown as a yellow cone, and the ground is depicted with green hills and a blue lake.

I. High altitude long endurance UAS conceptual study (2011-)



The image consists of two parts: an aerial view of the Fukushima Daiichi nuclear power plant on the left, and a small UAS in flight on the right.

II. Small UAS radiation monitoring for Fukushima (2011-)



The image shows a 3D diagram of a control station on the left and a UAS in flight on the right. The diagram is labeled 'image of our research' and includes the following components: 'Autonomous localization', 'Adhoc network', 'Exploration', 'Autonomous task apportionment', 'Control station', and 'Cooperation with a ground robot'.

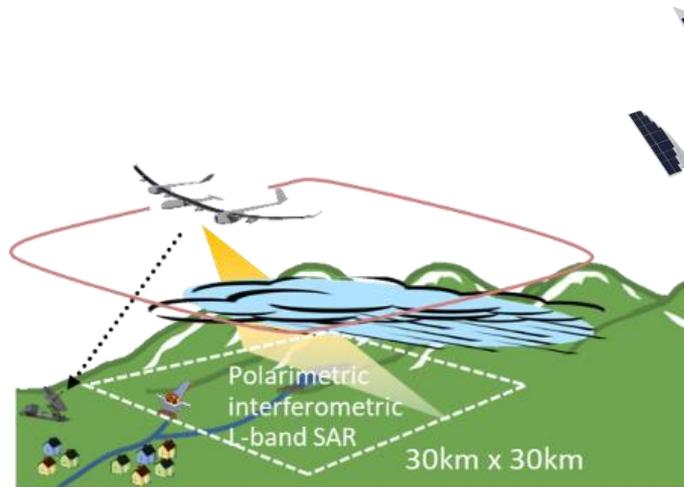
III. Multiple MAV corporative mission capability (2012-)



The image shows a satellite map on the left with a red circular area of interest and a UAS in flight on the right. The map displays various terrain features and a grid overlay.

IV. UAS traffic management and flight performance (2015-)

I. High Altitude Long Endurance UAS Conceptual Study



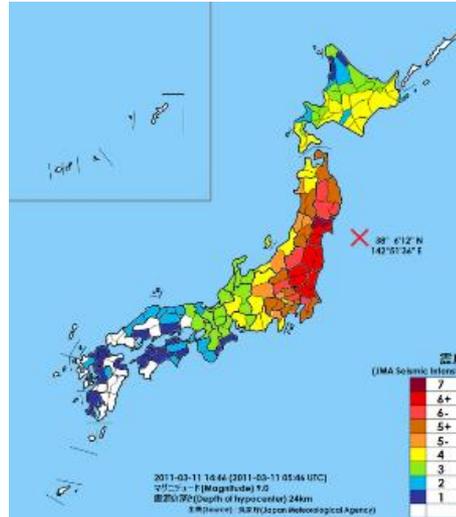
Background: Northeast Japan Huge Earthquake, 2011



Sendai airport¹⁾



Miyagino-ku, Sendai-city¹⁾

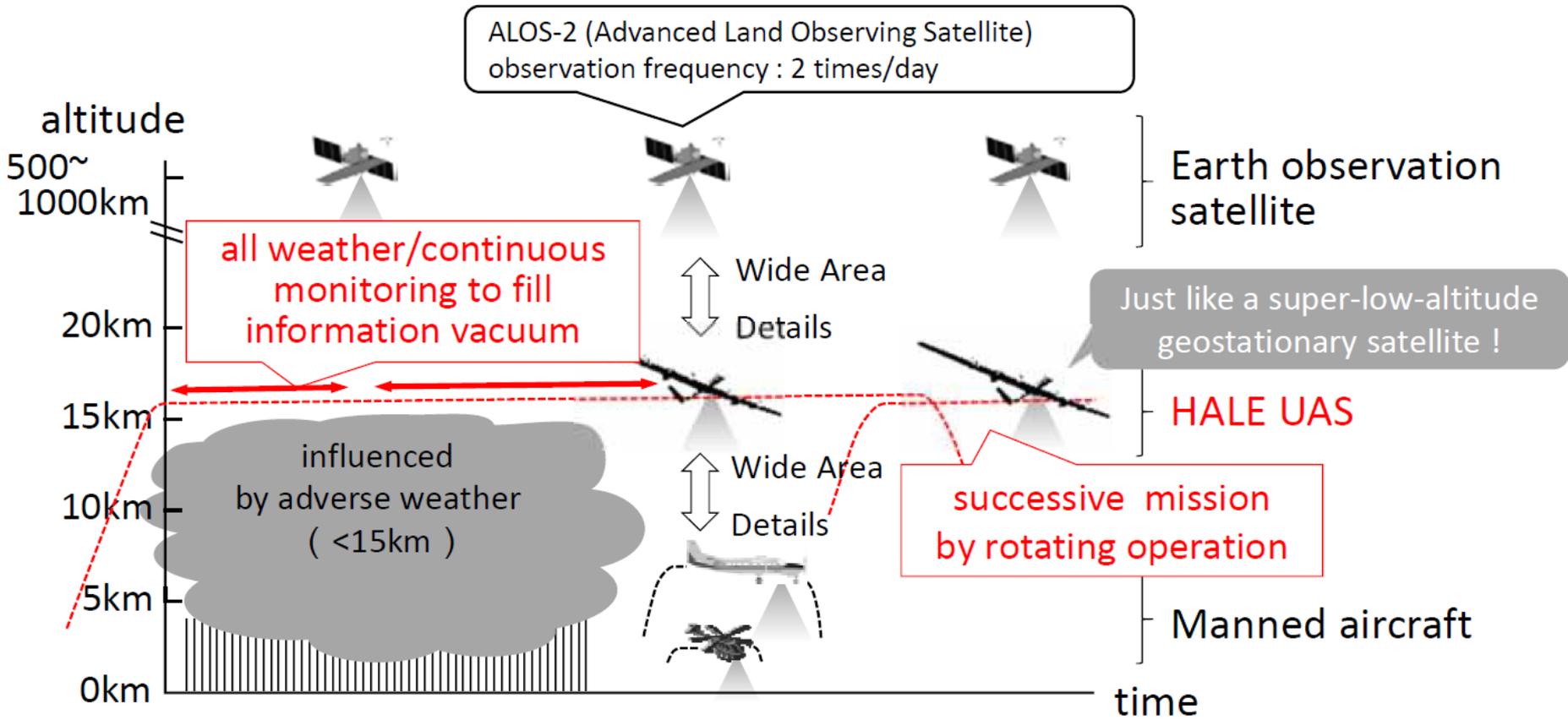


Miyagino-ku, Sendai-city¹⁾

1) <https://ja.wikipedia.org/wiki/%E6%9D%B1%E6%97%A5%E6%9C%AC%E5%A4%A7%E9%9C%87%E7%81%BD>

- Seismic intensity 9.0 (the Richter scale)
- Number of missing and dead: 18,446
- **Largest earthquake in Japan especially wide areas**
- **Huge tsunami, wide area**
- Resulting nuclear power plant accident
- **There is “small number” of actual use of UAS**

HALE UAS for a “continuous mission” platform



Target Flight Performance

Operating Altitude

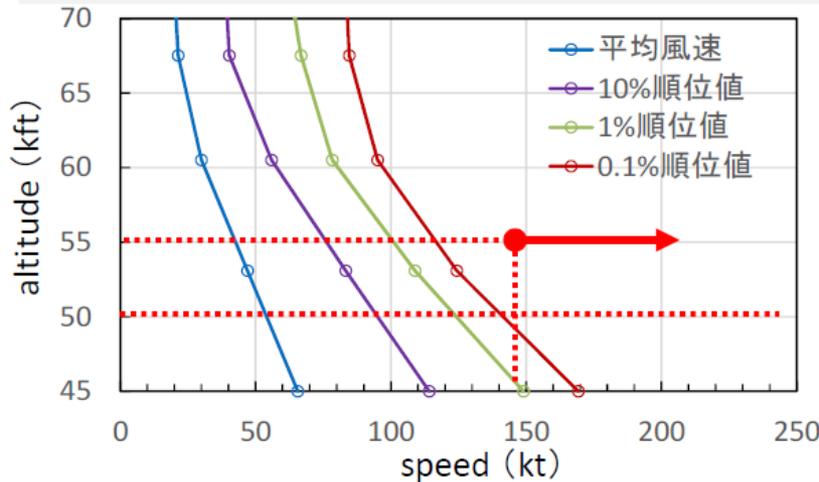
- above typhoon and cumulonimbus
- above the airspace of manned aircraft

55 - 60kft
(16.5 - 18km)

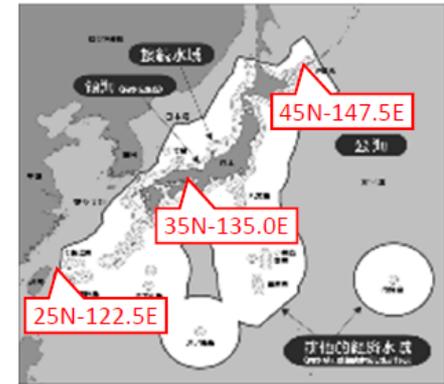
Cruise / Max. Speed

- more than 99.9% wind + 50kt* at operating altitude
- * ground speed needed for SAR

150kt (max. 210kt)
(280 - 390km/h)



Wind data :
NCEP Reanalysis data
at 3 locations, 2000-2009



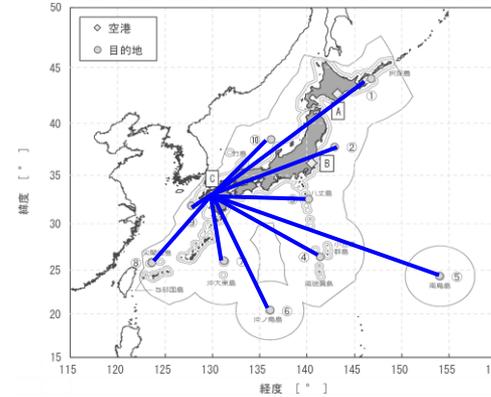
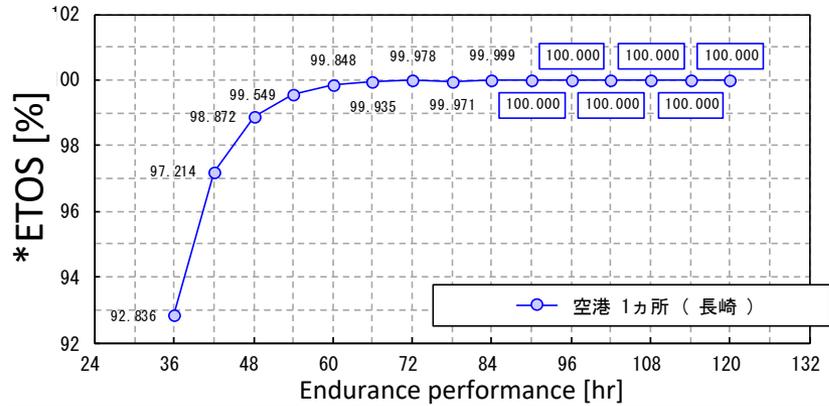
Endurance

- Effective Time On Station, ETOS \cong 100%

72hr

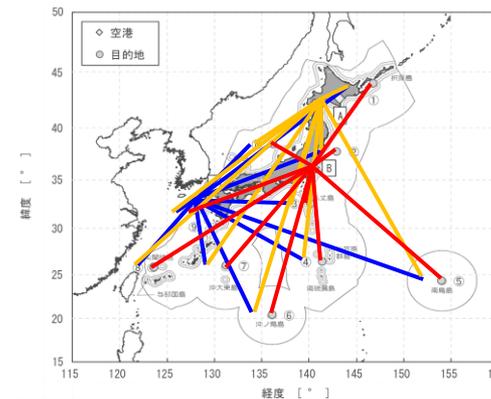
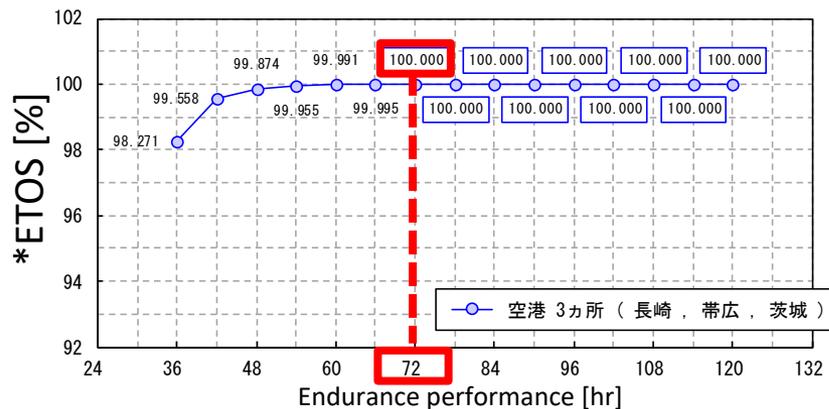
Operation analysis based on past airport weather data

Single airport operation



- Rotational operation
- Constraint of airport weather (cross wind and visibility)
- Cover exclusive economic zone of Japan

Multiple airports operation



- **72 hours endurance enables continuous operation capability**

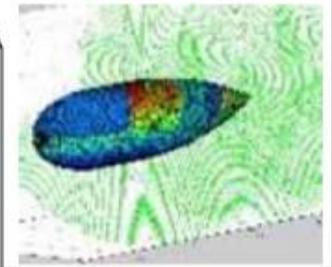
*ETOS: Effective Time On Station

Technological challenges to realize the flight performance

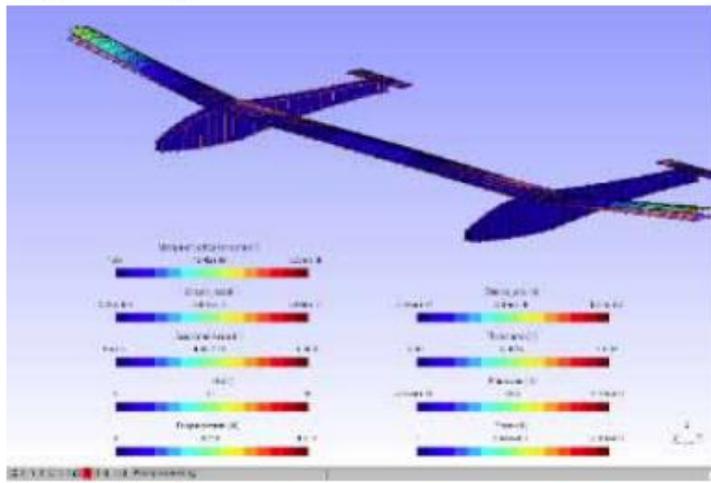
Wing span	46 m
MTOW	3.7 t
Payload	> 200 kg
Mission altitude	> 55kft (16.5 km)
Cruising speed	> 150 knots
Endurance	72 hours



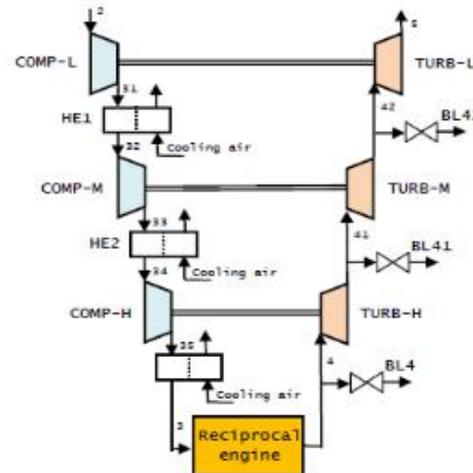
Low-Re propeller



Lightweight airframe structure

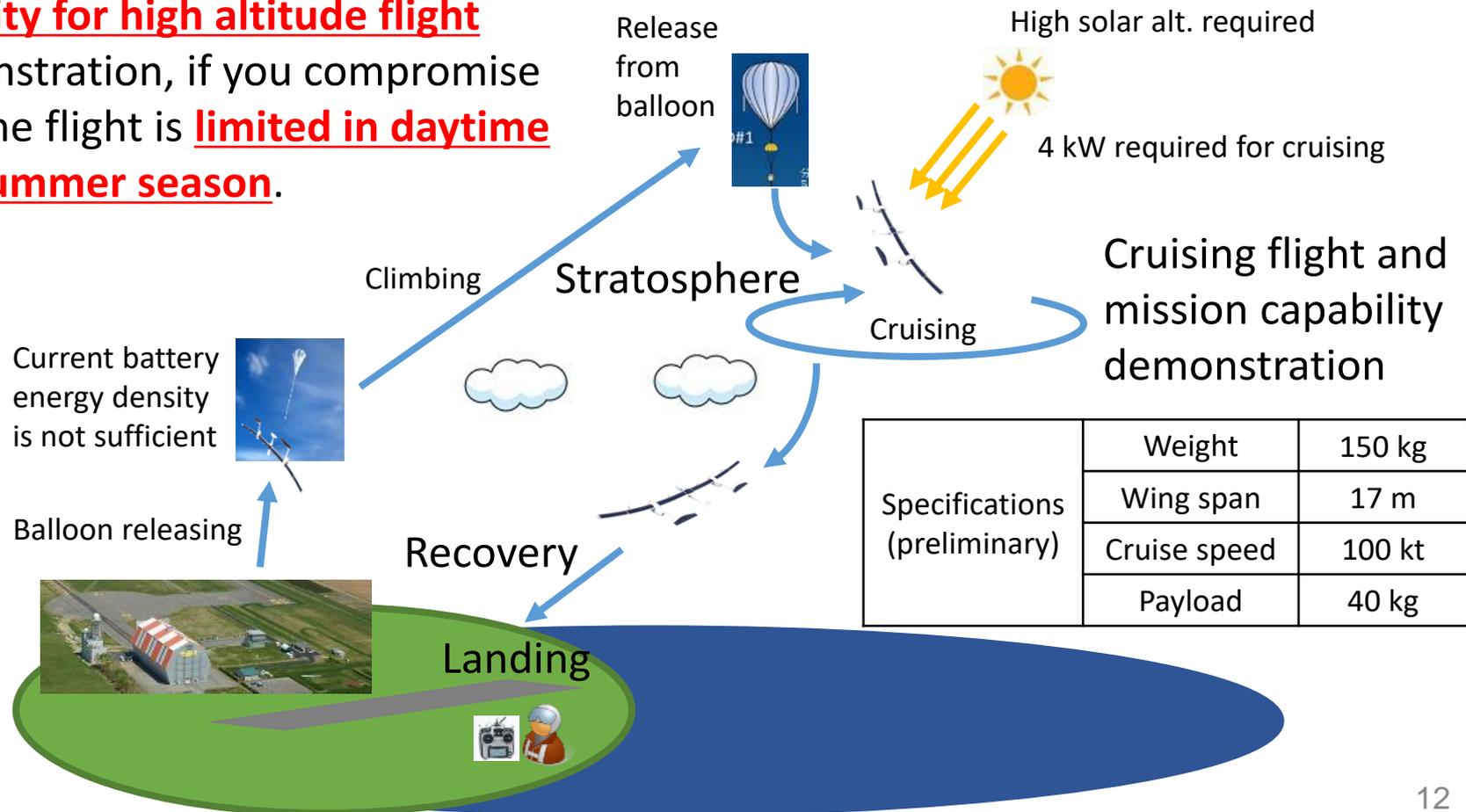


Multi-stage turbocharged diesel reciprocating engine

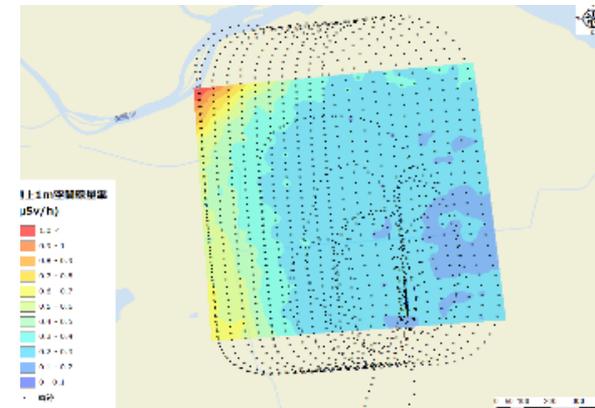


Small scaled HALE demonstration platform (preliminary)

Solar-powered high-speed airplane allows **large experimental-payload capacity for high altitude flight** demonstration, if you compromise that the flight is **limited in daytime and summer season**.



II. Fixed Wing UAS for Radiation Monitoring



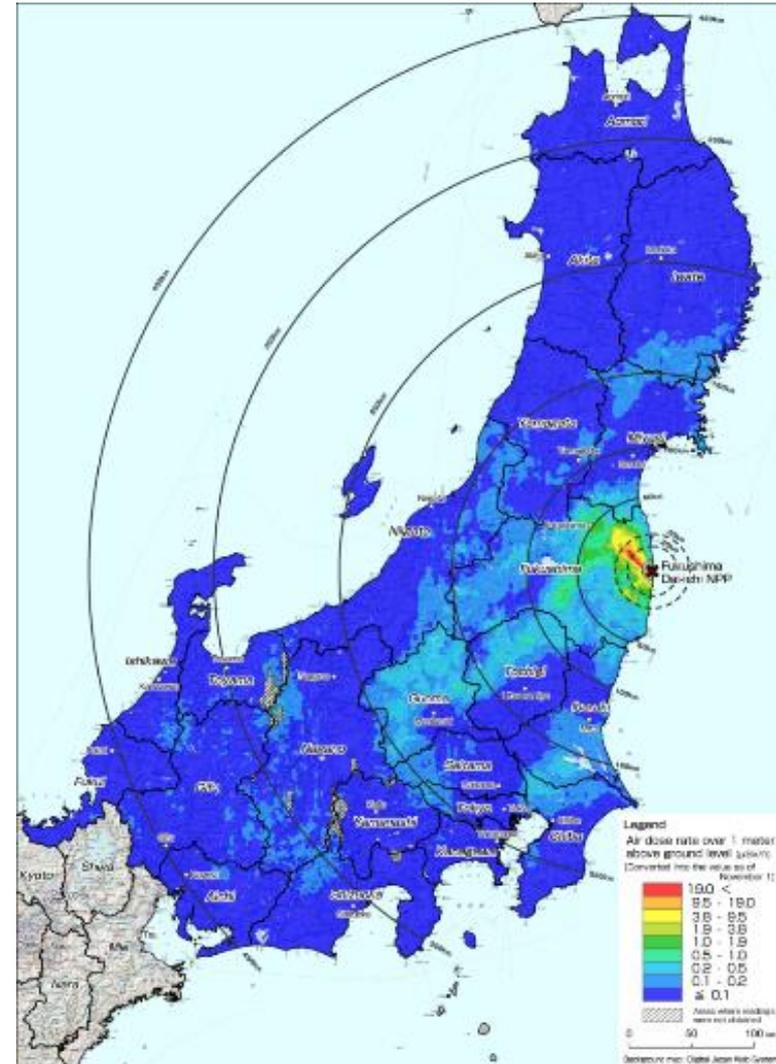
JAEA: Japan Atomic Energy Agency

UARMS: Unmanned Airplane for Radiation Monitoring System

Background: Nuclear Accident and Radiation



- ✓ Radiation monitoring using manned helicopters
- ✓ Continuous mission, not only once, but every year for monitor the variation
- ✓ A number of flights are required



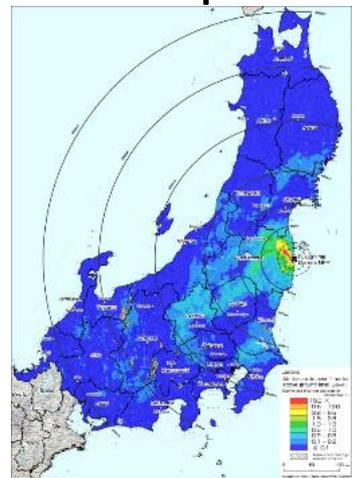
II. Fixed Wing UAS for Radiation Monitoring

UARMS project with the Japan Atomic Energy Agency (JAEA)

Range	Wide >100 km	Middle >10 km	Local ~1 km	Site ~100 m
Platform	Manned helicopter	Fixed wing UAS	Rotary wing UAS	Small UAS
Altitude	~300 m	~150 m	~50 m	~10 m



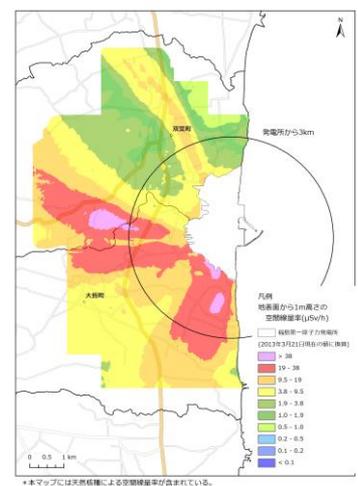
Wide area map



Middle area and rapid response in emergency (ex. forest fire)



1) <http://www.japantimes.co.jp/wp-content/uploads/2017/05/n-fukufire-a-20170512-870x650.jpg>



JAEA developing

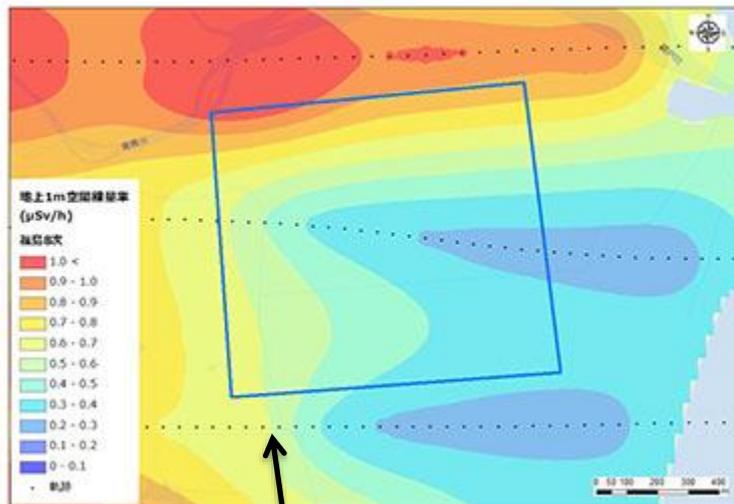
Local area detailed map 15

Radiation map obtained by fixed wing small UAS

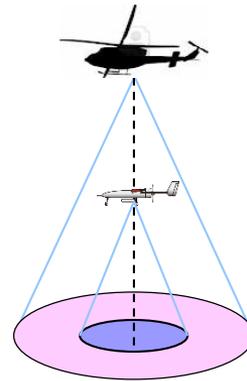
- ✓ World first application of fixed wing UAS for radiation map construction in 2014
- ✓ UAS has large benefit compared to manned aircraft
 - ✓ Low altitude, repeatable flight path, precise terrain following, etc.
- ✓ Miniaturization of precision radio detector is also key factor



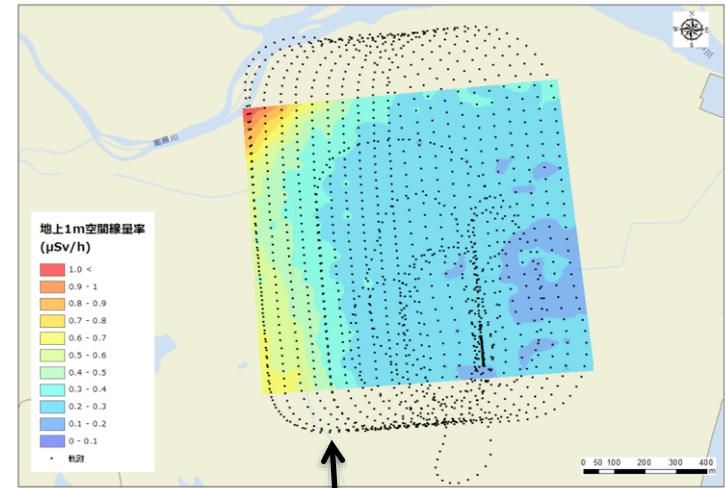
Manned helicopter = AGL 300 m



Dot line: low density and flight patterns



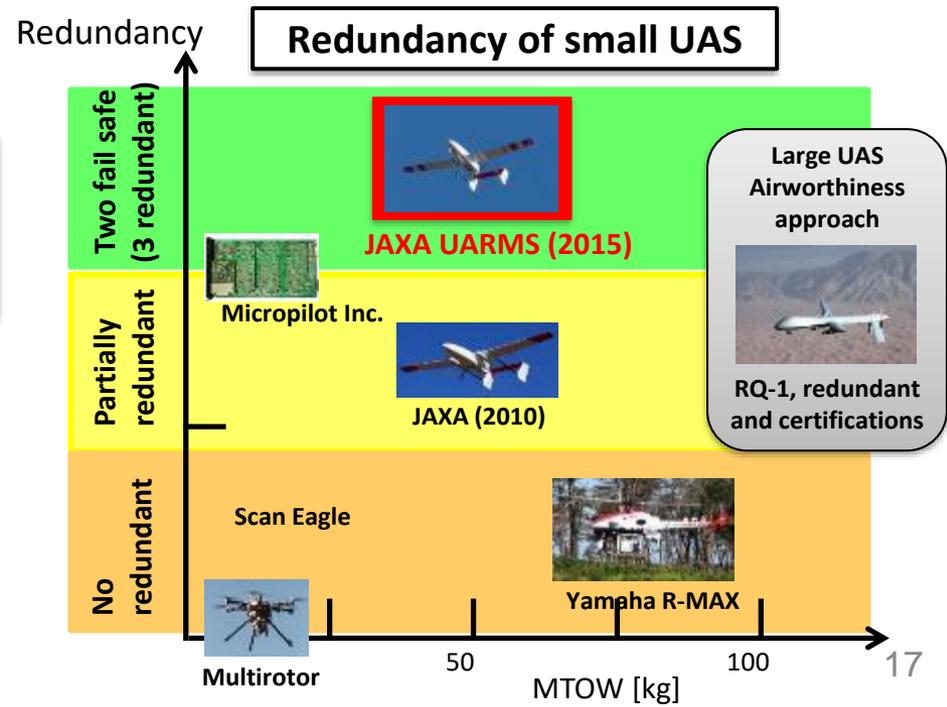
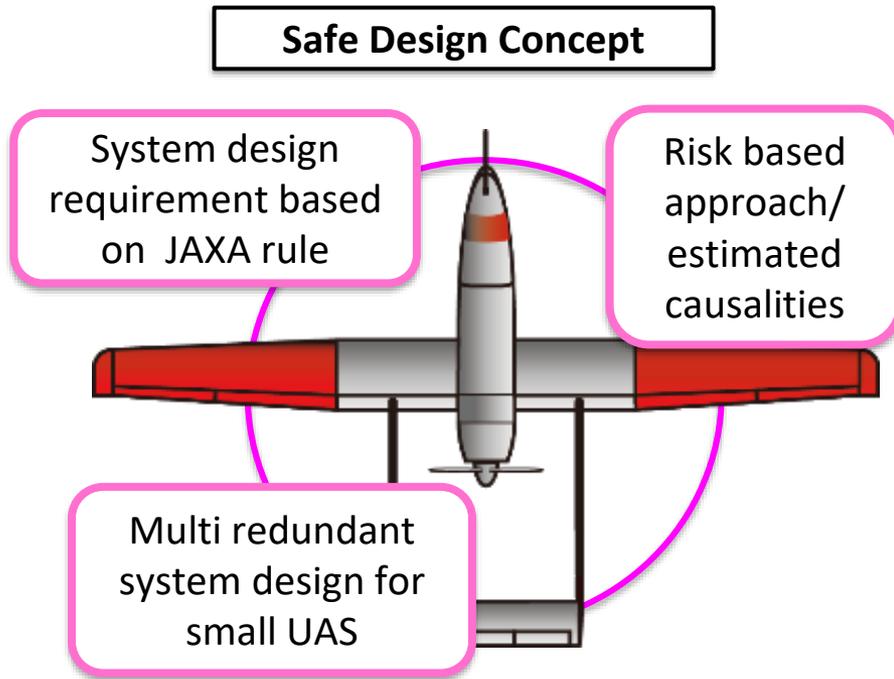
UARMS < AGL 150 m



Dot line: high density flight patterns

Safe Design Concept

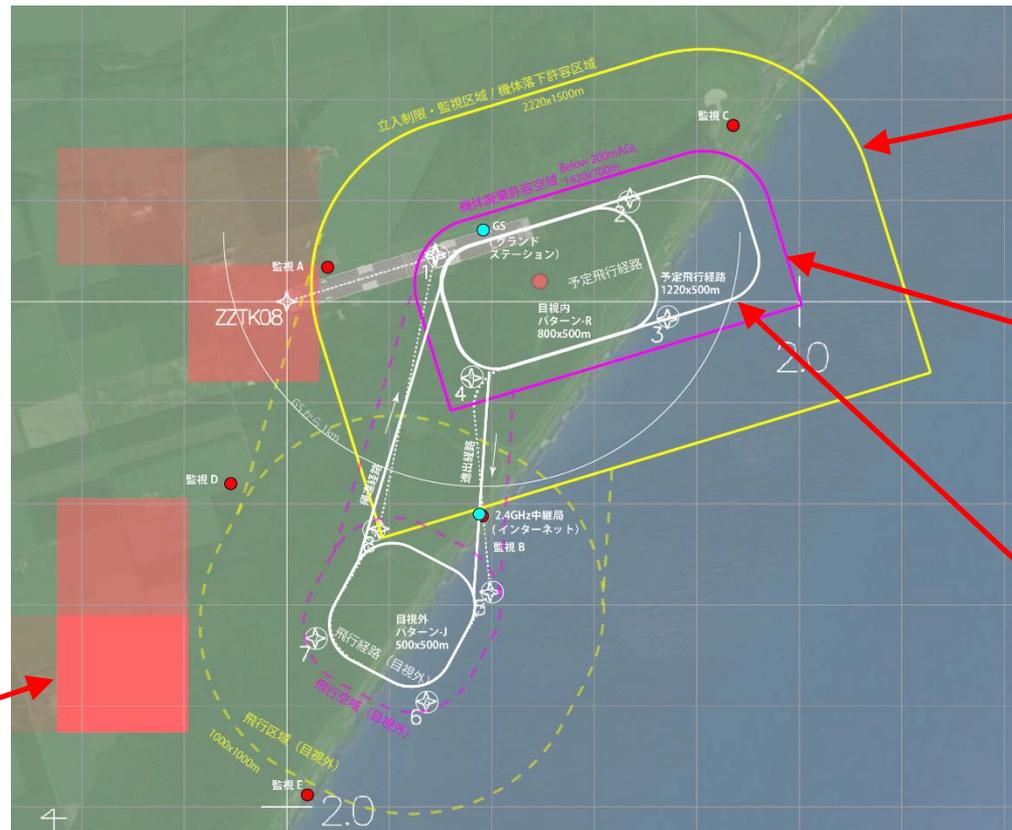
- ✓ Multi redundant design, two fail safe to catastrophic hazard
- ✓ Design requirement defined in JAXA rule
- ✓ Reliability evaluations based on operational data
- ✓ Beyond Visual line of Sight (BVLOS) operation planned



II. Fixed Wing UAS for Radiation Monitoring

Operation example: BVLOS flight demonstration

- ✓ JAXA Taiki Aerospace Research Field
- ✓ Safety management based on risk: “reliability of UAS” x “population density”
(maximum population density = 0.4/km²)



Ground safety control area boundary

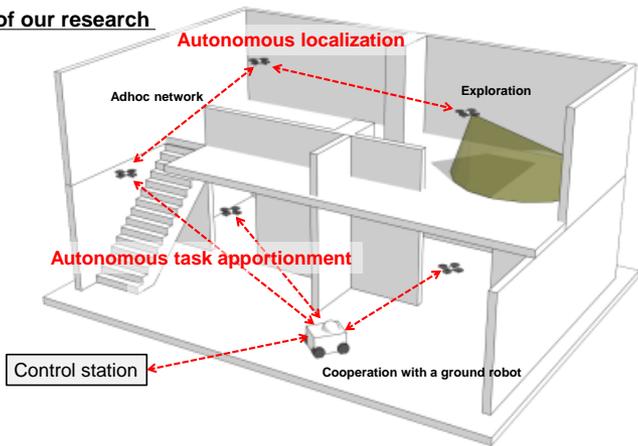
Flight area boundary

Planned flight pass

Population density (500 m mesh)

III. Multiple MAV Corporative Mission Capability

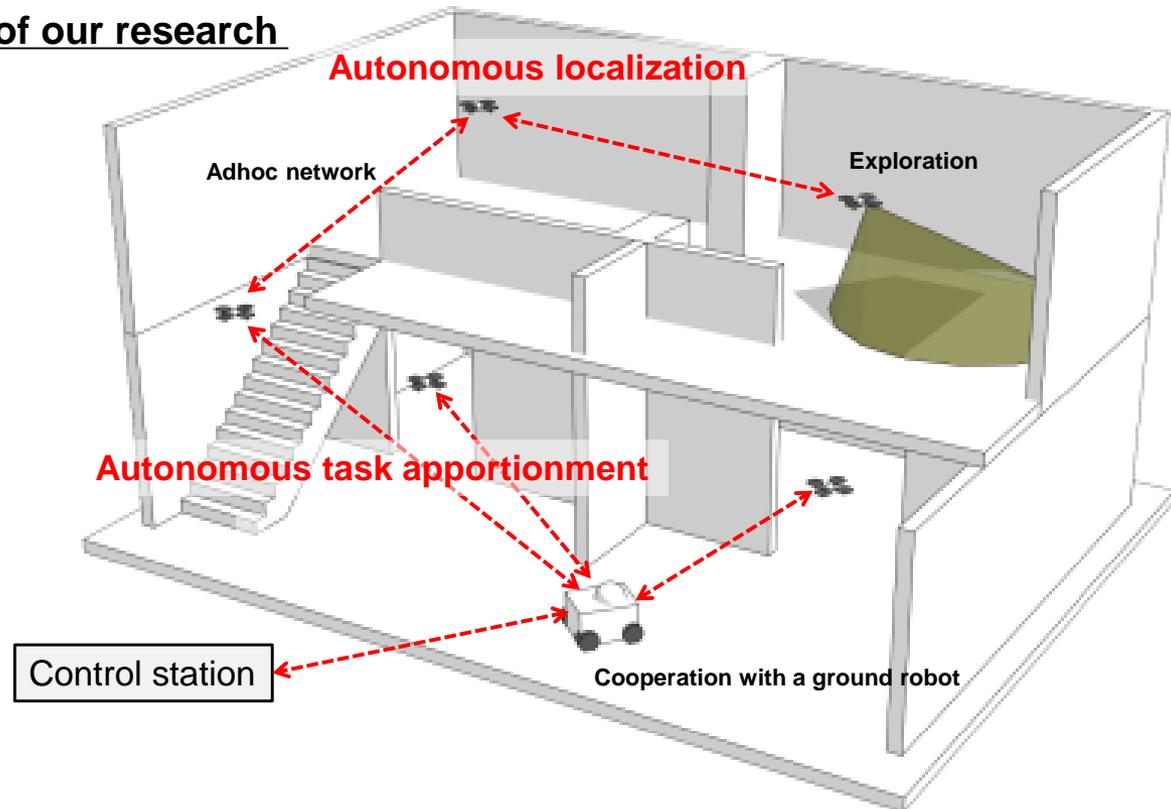
Image of our research



Technical challenges

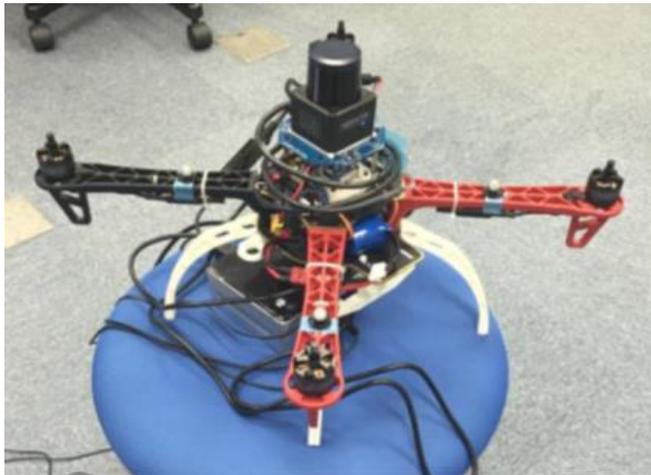
- ✓ GPS denied navigation (multiple SLAM, advanced optic flow)
- ✓ Cooperative motion planning algorithms: simple and robust rule-based, greedy-approach, and distributed algorithm
- ✓ Safe flight platform for indoor environment

Image of our research

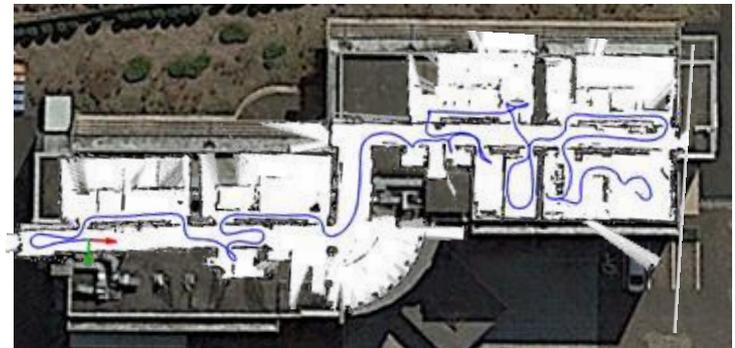
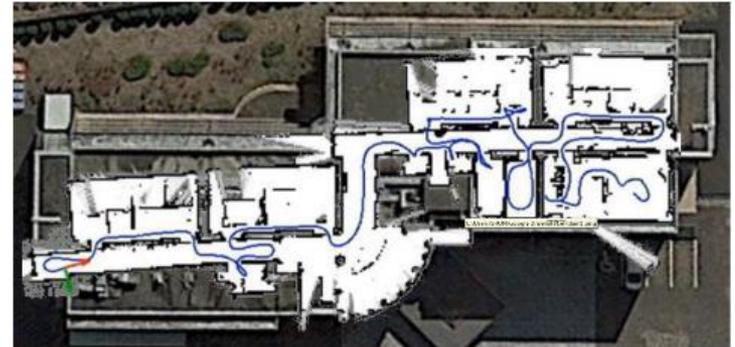


GPS Denied Navigation

- ✓ SLAM: simultaneous localization and mapping
- ✓ Developed in ground robotics area and recently applied for MAV
- ✓ Challenge: multiple flying vehicle data real-time integration



Small Lidar equipped MAV for SLAM research



SLAM occupancy grid map overlaid on actual building
Distortions are issue that is currently addressed via loop closing technique

Cooperative motion planning algorithms

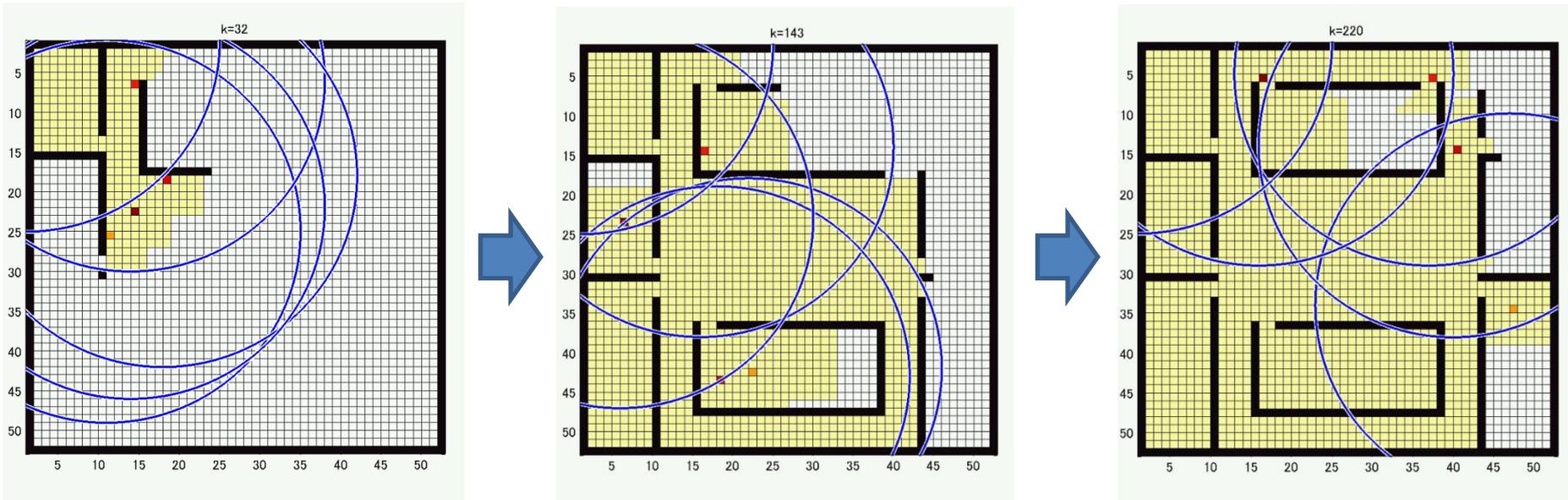
- ✓ Simple and robust rule-based
- ✓ Greedy-approach
- ✓ Distributed algorithm for multipole MAVs operation

Real time HILS



Simulation of the algorithm

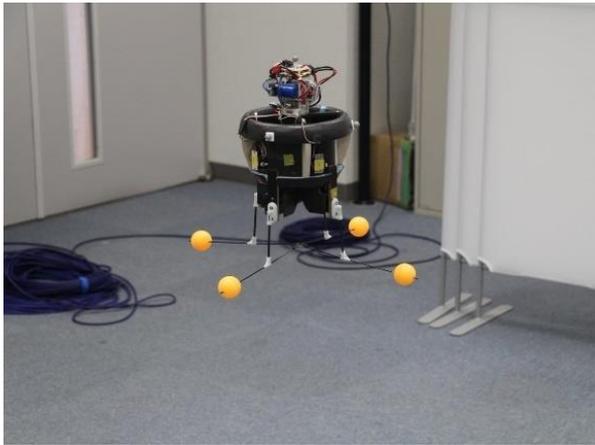
Red dot: MAV, **Blue circle:** communication relay constraint



More safe platform technologies: ducted-fan type MAV

Palm-sized single ducted fan flying robot has a high level of safety because of the shrouded propeller, which is harmless to ground objects and people.

The undesired noise because of the high speed rotation and interference of duct structures will be the biggest problem and the research task.



15 cm-duct-diameter
ducted fan MAV

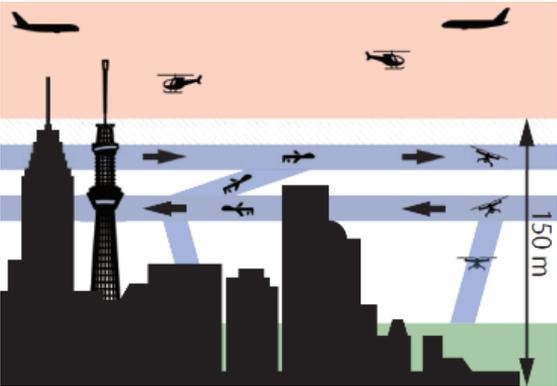


10 cm-duct-diameter
ducted fan MAV



Smaller **7.5 cm**-duct-
diameter ducted fan MAV

IV. UAS Traffic Management and Flight Performance



Background: BVLOS flight for nuclear reconnaissance



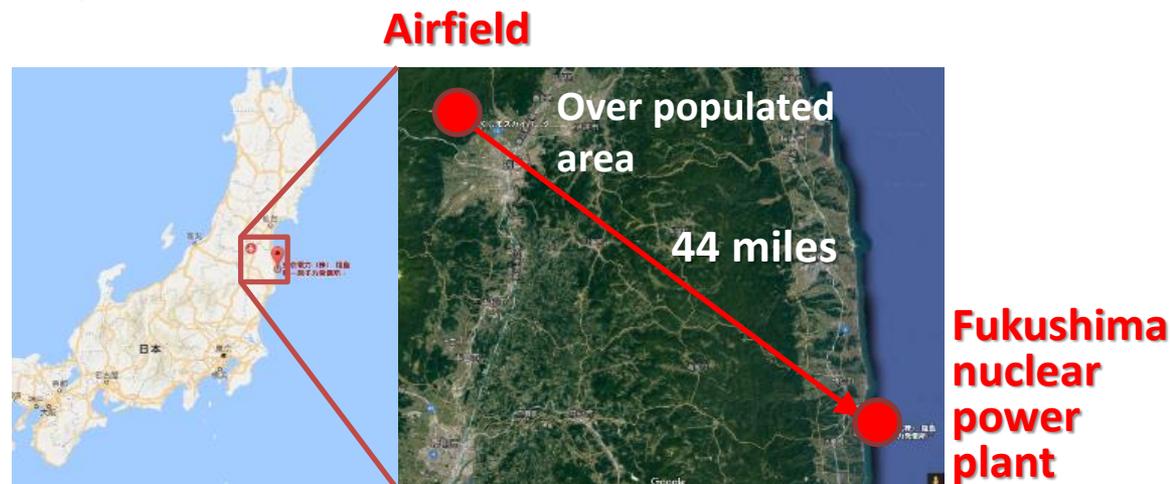
Fuji-imvac Type B¹⁾



Fukushima dai-ichi nuclear power plant¹⁾
03/20/2011 (9 days after the accident)

1) <http://www.yamazaki-k.co.jp/airphoto/concept.html>

- BVLOS (beyond visual line of sight) operation
- No long distance command and control links
- Flight over populated area at altitude of 1,200 m, 2 hour flight including 1 hour stay over the power plant
- Airspace segregated by civil aviation authority
- Safe and regular operation technologies are required



Background: more sUAS in recent disaster, now



Heavy rain fall and landslide at Hiroshima, 2014¹⁾



Kumamoto earthquake 2016²⁾



Media drone “chased” disaster relief helicopter flying low altitude³⁾



Geographical Survey



Infrastructure damage assessment by company

- Small UAS are now more popular and many VLOS operations
- Small UAS were flown by governmental institute, universities, companies, NPOs, news medias
- **Conflict resolution will be required in future disaster**

1) <https://blogs.yahoo.co.jp/y294maself/34523293.html>
2) <https://news.yahoo.co.jp/story/146>
3) <https://weathernews.jp/soramagazine/201705/08/>

UTM concept proposed by NASA



NASA UTM

<https://utm.arc.nasa.gov/documents.shtml>

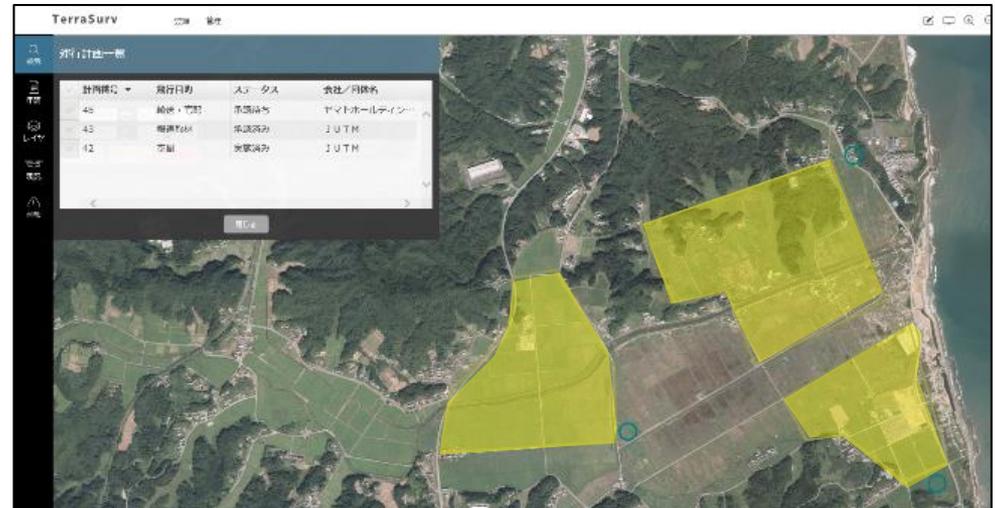
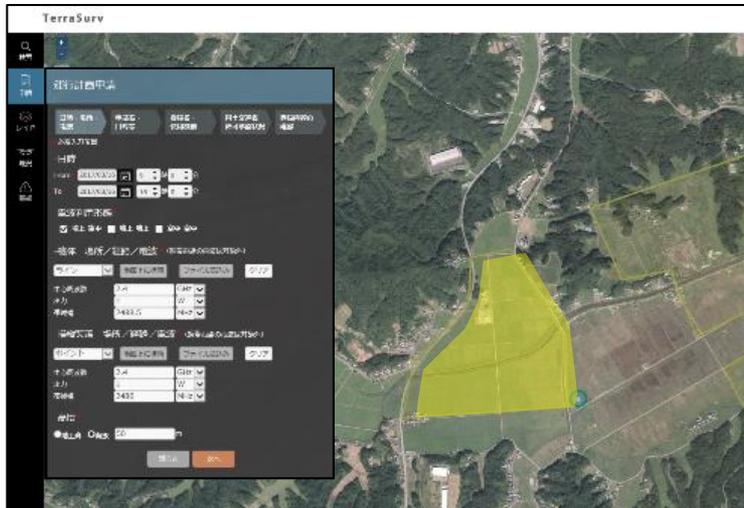
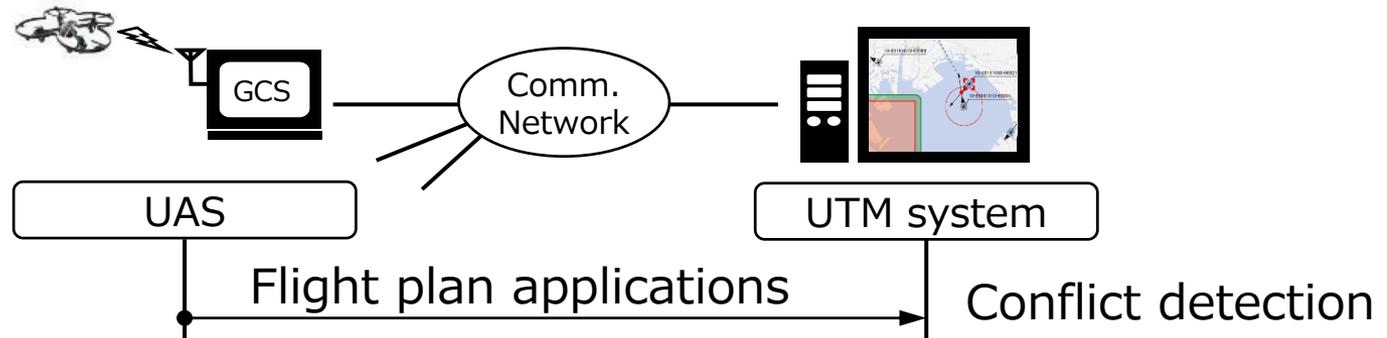


ICAO

*“Currently, there is no established **infrastructure to enable and safely manage the widespread use of low-altitude airspace and UAS operations**, regardless of the type of UAS. A UAS traffic management (UTM) system for low-altitude airspace may be needed, perhaps leveraging concepts from the system of roads, lanes, stop signs, rules and lights that govern vehicles on the ground today, whether the vehicles are driven by humans or are automated.”*

(cited from NASA UTM website)

UTM basic research and experiment in Japan



Flight performance study: gust tolerance of small UAS



Gusty wind ??? 1)

- Wind tolerance is one of the most important technical challenges for the practical use of drones
 - sUAS's relatively light weight and low-altitude operation
 - Too high "drone user's requirement" of position control accuracy even in gusty wind
- For beyond visual line of sight (BVLOS) operations, guidance accuracy degradation due to the existence of gust wind is an important factor

1) <http://www.triple.com.au/melbourne/news/blog/2013/7/wind-to-smash-victoria/>

Gust wind tunnel flight for direct gust response evaluation



2x2 m low speed WT with gust generator

Using a gust wind tunnel facilities and a motion capture based precise flight control system, direct gust tolerance performance evaluation method was constructed.



Actual flight in gust wind tunnel



Precise navigation and guidance using a motion capture technology

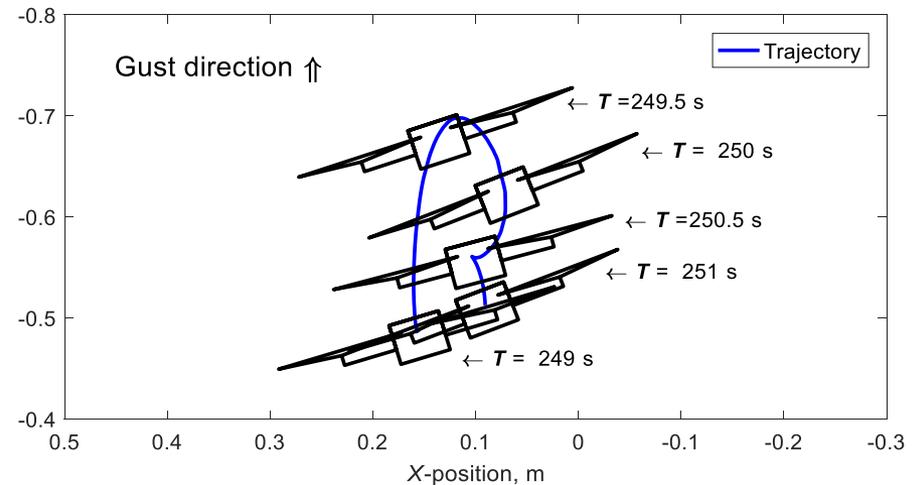
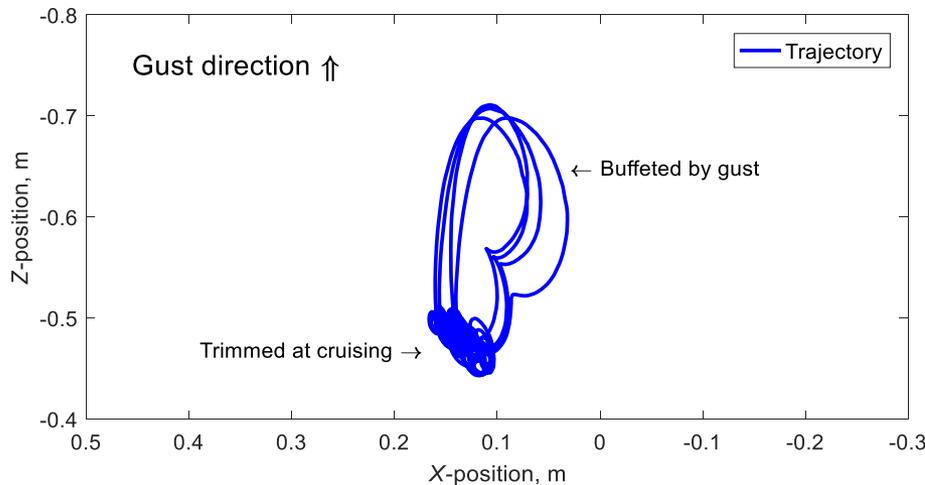
Advantages:

- ✓ Free flight without any aerodynamic interference
- ✓ Reproducibility and quantitiveness

Limitations:

- ✓ The wind tunnel can generate only brown-up and down gust, however, cross wind gust is also important

“1-cos Type” Isolated Gust Response (1Hz, 10 m/s)



- Reproducible gust response trajectories were obtained

- Pitch attitude was not much affected by the blown-up gust, which is different from those of head or tail-wind gust response in another experiment

Disaster Risk Reduction using UAS

I. High altitude long endurance UAS conceptual study (2011-)

II. Small UAS radiation monitoring for Fukushima (2011-)

III. Multiple MAV corporative mission capability (2012-)

IV. UAS traffic management and flight performance (2015-)

Thank you for your kind attention!
Please do not hesitate to visit Japan!!

JAPAN

JNTO Japan National Tourism Organization



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