





Effects of Wing Sweep on In-Flight Boundary-Layer Transition for a Laminar Flow Wing at Mach Numbers From 0.60 to 0.79

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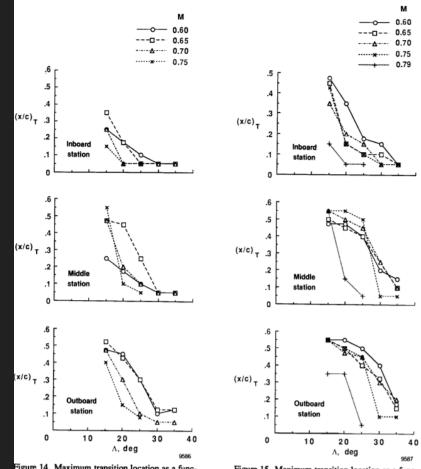


Figure 14. Maximum transition location as a function of sweep, hp = 10,000 ft.

Figure 15. Maximum transition location as a function of sweep, hp = 20,000 ft.





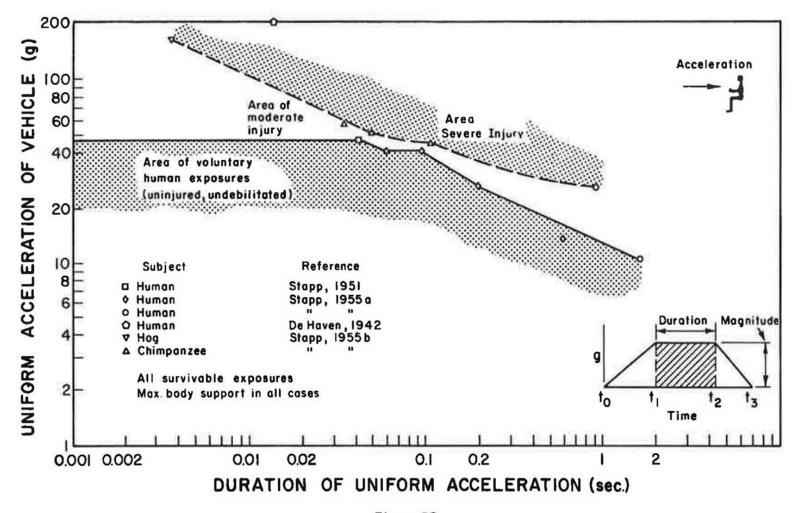


Figure 23.

